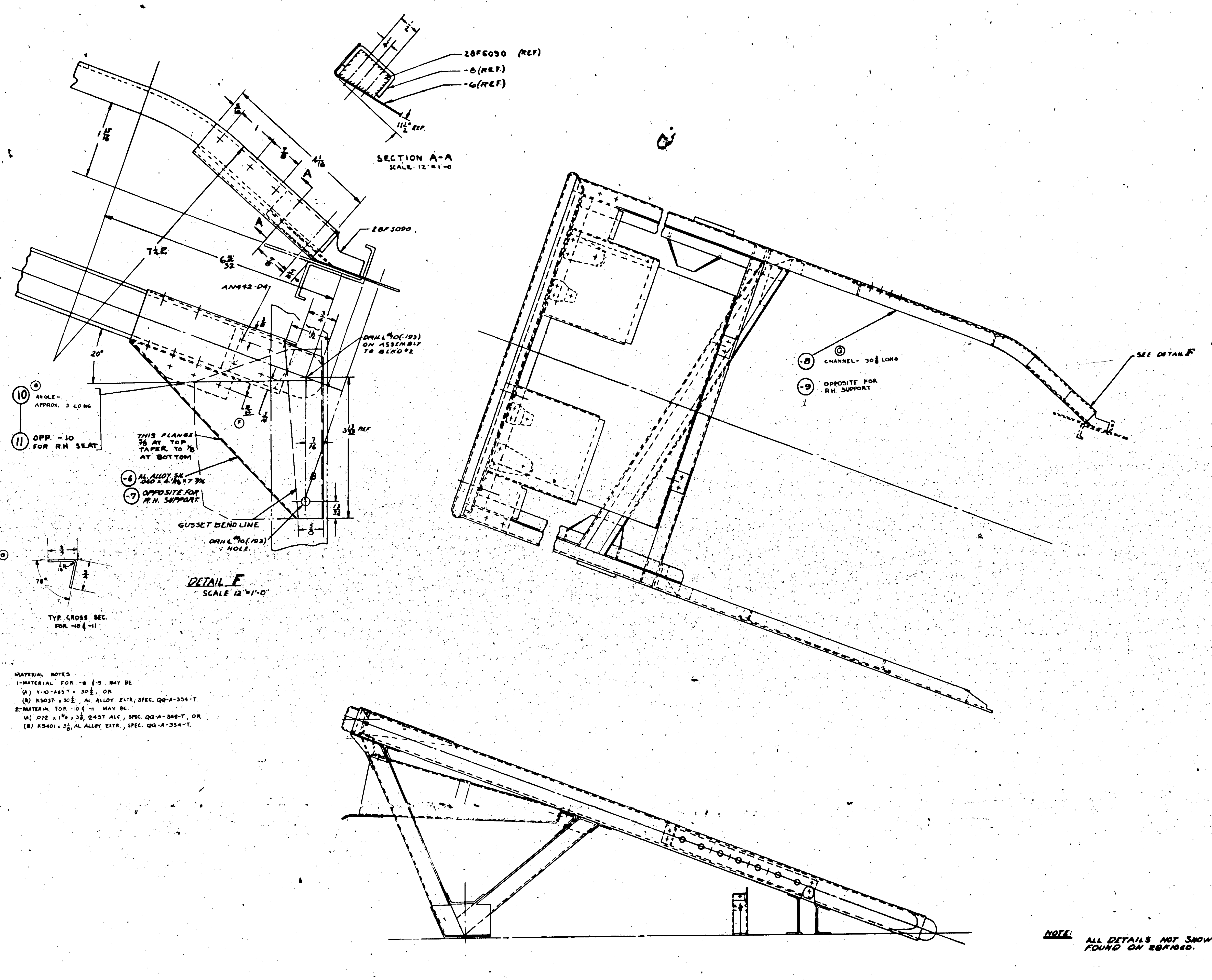


MATERIAL NOTES

1-MATERIAL FOR -8 & -9 MAY BE
(A) Y-10-A85T x 30 1/2, OR
(B) K5037 x 30 1/2, AL. ALLOY EXTR., SPEC. QQ-A-354-T

2-MATERIAL FOR -10 & -11 MAY BE
(A) .072 x 1 1/8 x 3 3/8, 24ST ALC., SPEC. QQ-A-362-T, OR
(B) K5401 x 3 1/8, AL. ALLOY EXTR., SPEC. QQ-A-354-T.

NOTE: ALL DETAILS NOT SHOWN, FOUND ON 28F1060.

2- ALL RIVETS TO BE AN442-D3 UNLESS OTHERWISE NOTED.
1- FINISH - ANODIZE ALL AL. ALLOY PARTS CADMIUM PLATE-37
P-AP BEFORE ASSEMBLING 22F055-9 (ON 28F285)

CNOA (S) 464
C 91876 #495-499 & 510 & ON
28F1060-1 R.H. SUPPORT OPPOSITE.
28F5060 L.H. SUPPORT AS SHOWN.
C91876 #1-494 & 500-509
C88476 & C#77713-#4 & ON ONLY
C#70496, #168-200 & 77713 #1-3

FINISH CODE	
Z	ANODIC TREATMENT
P	P-27 PRIMER
AP	ALUMINIZED PRIMER
C	CADMIUM PLATE

CONSOLIDATED AIRCRAFT CORP.

SUPPORT-FURNISHINGS PILOTS SEAT

J 28F5060

R

H 319